A Highly Scalable, Low Latency, and Deterministic Flight Control System for UAS Estimation and Flight Controls Research

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University of Minnesota UAS Research Labs

- Focus on flight data to validate research
- Navigation, guidance, and flight control research and education
 - Safety and efficiency
- Flight research and test techniques
 - Test design, uncertainty estimation, system id and parameter estimation
- Applications of UAS
 - Agriculture, invasive species, and surveys





Motivation



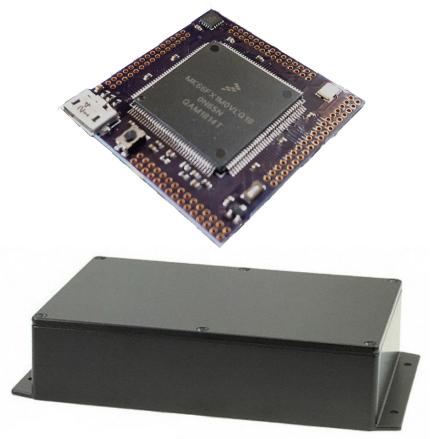


- Scalability
 - Widely varying system requirements
 - UAS size, type, sensors, actuators
- Low latency and determinism
 - Flexible aircraft
- Rapid flight validation
 - Limited resources, fast pace of testing
- License structure
 - Open source for publication
 - Ability to work with businesses and research institutions

Bolder Flight Systems

- UAS Labs spinoff
- Platform to rapidly research, develop, and commercialize aeronautics technologies
- Develop, manufacture, and commercialize:
 - Hobbyist and educational breakout boards
 - Flight control systems
 - Research aircraft
 - Proprietary derivatives
- Engineering, flight test, and analysis services
- Designed and manufactured in Oregon





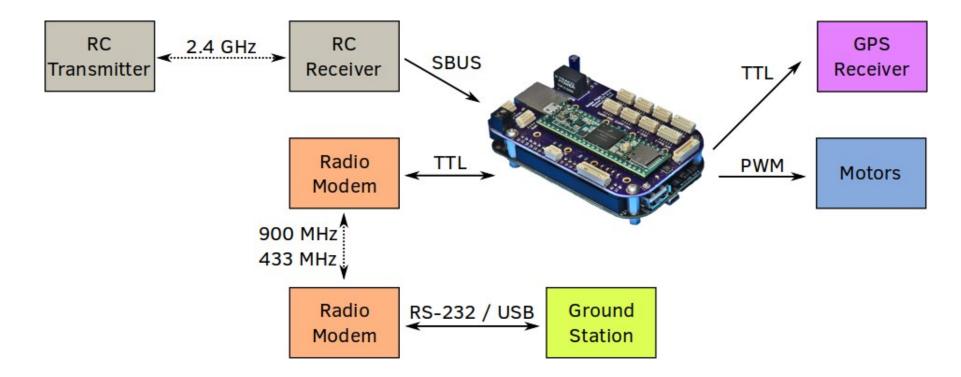
System Overview





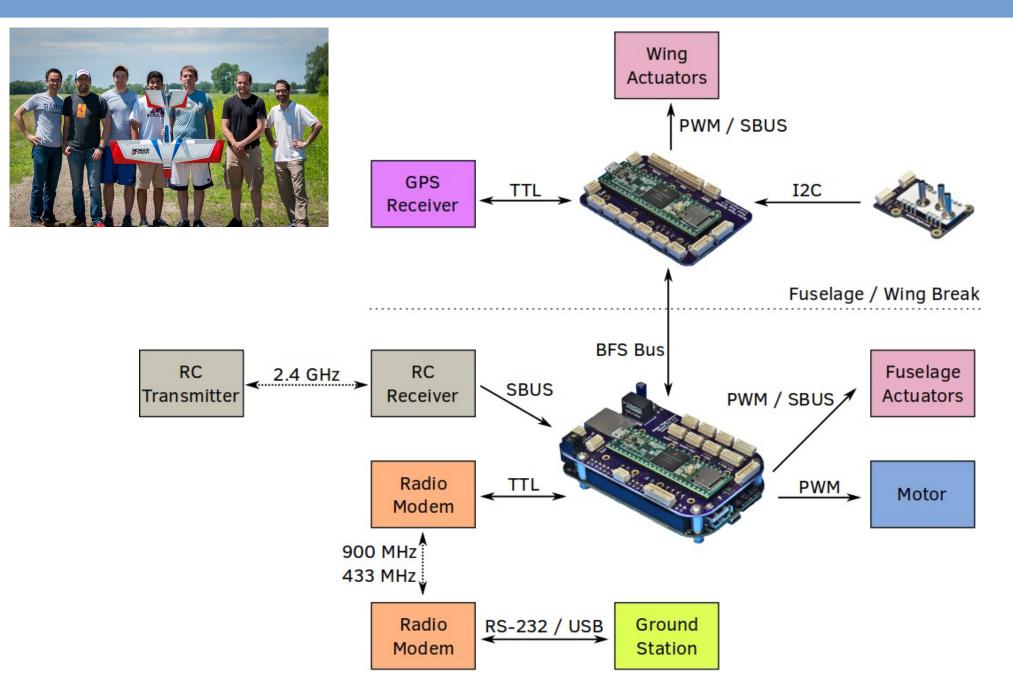
- Flight management unit
 - Systems, sensor, and actuator management
 - Hard real time
 - Integrated voltage regulation and sensors
 - SBUS, PWM, I2C, SPI, UART
 - Baseline sensor processing, estimation, and control laws
- SOC
 - Research / application interface
- Node
 - Sensor and actuator expansion

Functional Block Diagram

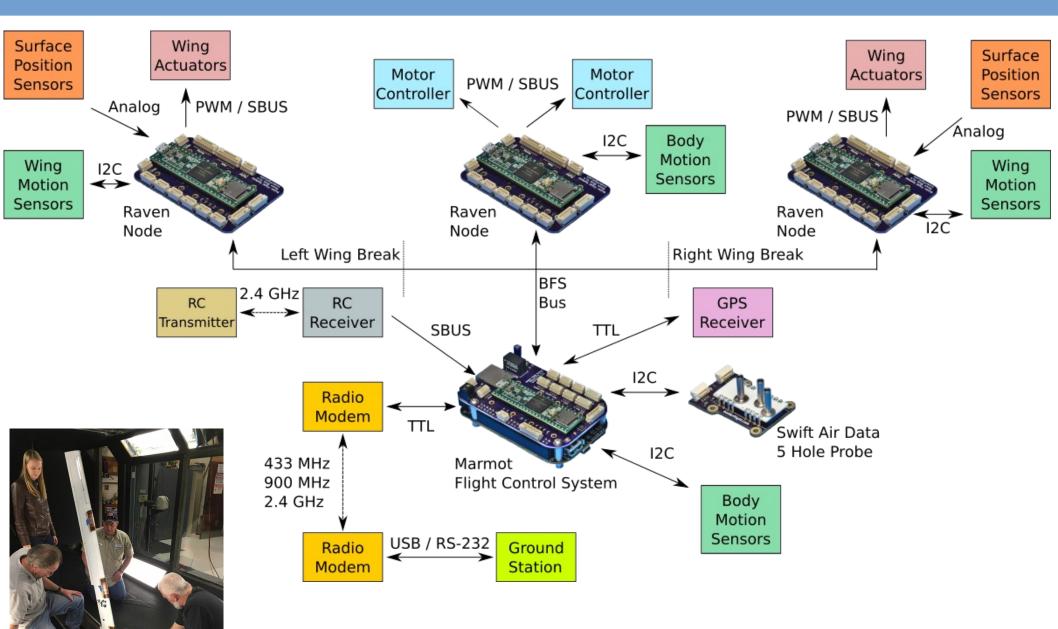




Functional Block Diagram

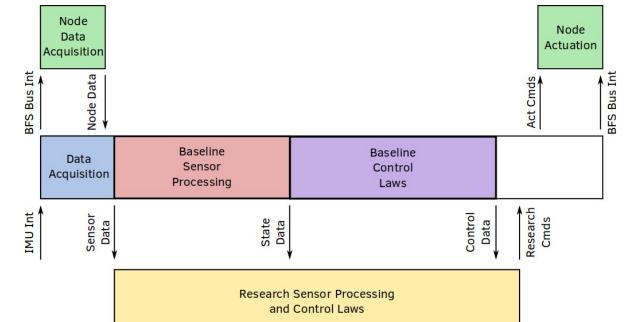


Functional Block Diagram



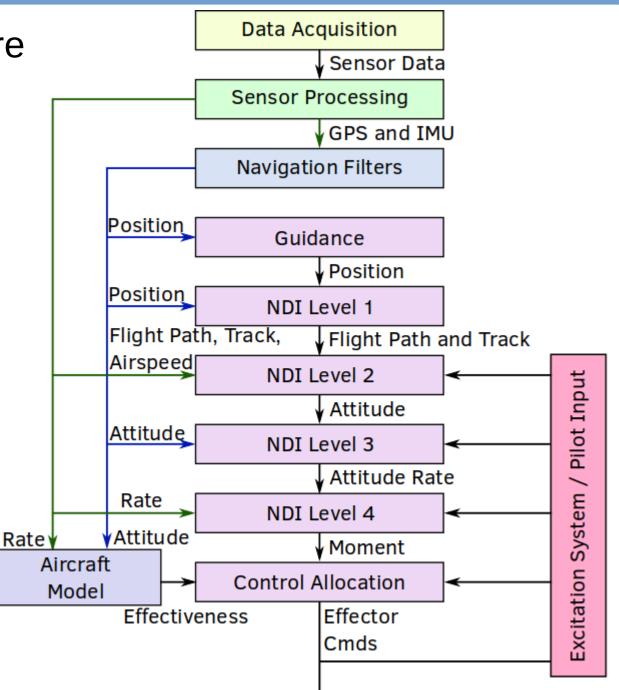
Hardware

- FMU / Node
 - 240 MHz Cortex M4, single precision hardware floating point
 - Hardware crypto unit
- SOC
 - 1 GHz Cortex A8
 - High bandwidth connection to FMU (3 Mb/s)
- BFS Bus
 - High bandwidth connection from FMU to Nodes (2 Mb/s)
 - Digital lines to sync data acquisition and actuation



Software Approach

- Cascaded flight software
 architecture
 - Logical grouping
 - Flexible reference command source
 - Modular configuration



Software Environment

- Development
 - C 11 / C++ 17
 - General use sensor drivers, communication, and estimation algorithms
 - Widely used for other projects and platforms
 - Common utility classes
 - Eigen matrix math library
 - Timing, hardware drivers
- Protocol Buffers
 - Flexible research / application interface
 - Broad language support
 - Easy to add new object and message definitions
- Cmake build tools



MPU-9250 Sensor Driver

SBUS Receiver and Actuator Driver



Configuration Overlay

- Configurable software
 - Objects
 - Hash tables
 - Functions, managers, and factories

{ "Type": "PID2", "Reference": "/Control/refV_ms", "Feedback": "/Sensor-Processing/vIAS_ms", "Output": "cmdMotor_nd", "Sample-Time": 0.02, "Gains": {"Proportional": 0.200, "Integral": 0.075}, "Limits": {"Upper": 1, "Lower": 0} }

"SOS": { "Type": "MultiSine", "Duration": 20.0, "Frequency": [0.9425, 1.5708, 2.5133, 4.0841, 6.5973, 10.6814, 17.2788], "Phase": [0.0, 0.0, 0.0, 0.0, 0.0, 0.0, 0.0], "Amplitude": [1.0, -0.6, 0.375, -0.2308, 0.1429, -0.0882, 0.0545]},

"Test-Points": [
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Case Study

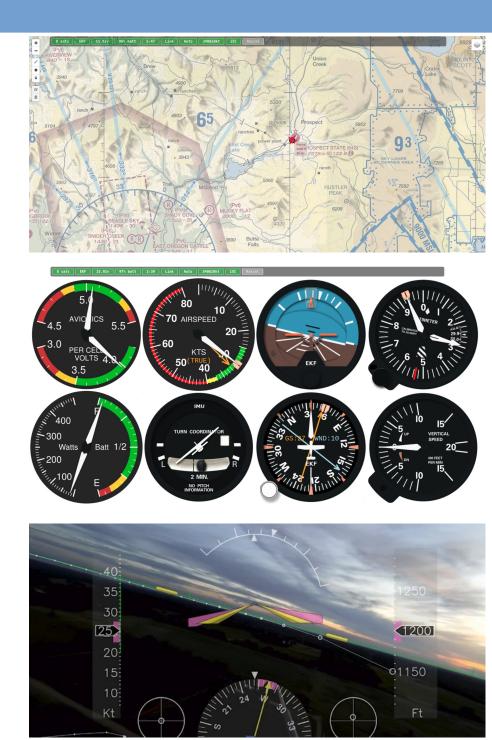
- System id and handling qualities research
- Ultra Stick 120 fixed wing UAS and a quad rotor UAS
- Similar excitations and maneuvers
- Very different vehicle, sensor, and actuator configuration
- Able to limit the configuration differences to the sensor, actuator, and inner loop control laws
- Reuses outer loop, excitation system, and test point definitions
- Initial flights on Ultra Stick 120 completed, currently flight validating quad rotor inner loop control laws





Future Work

- Updating APIs and research
 / application interface
 - Much easier integration of new sensors and software
- Graphic configurator
- Ground station
- MATLAB and Simulink
 integration
- Autonomous Flight Development
 - Model updating, control law tuning, flight envelope clearance



Questions

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